

# Guideline development for offshore structure vibration analysis

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Table 1. Synopsis

Section	Performed work	
<i>1. Special studies</i>	Mesh convergence: <ul style="list-style-type: none"> <li>• beam;</li> <li>• cylindrical shell;</li> <li>• plate;</li> <li>• stiffened panel.</li> </ul>	Parameter sensitivity: <ul style="list-style-type: none"> <li>• material;</li> <li>• 1D vs. 2D vs. 3D finite elms.;</li> <li>• effective modal mass;</li> <li>• added mass.</li> </ul>
<i>2. Simple case</i>	The global response of an electrical motor assembly.	The design of a vibration isolation system.
<i>3. Complex case</i>	The local response of a retractable thruster.	

- Service limit state;
- High structural complexity;
- Modal analysis.

### Working Hypotheses:

- Steady state response;
- No stress analysis;
- Periodic loading.

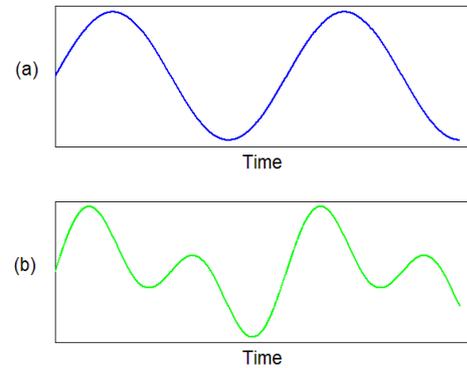


Figure 1. Characteristics and sources of periodic loading:  
(a) simple; (b) complex

# BASIC SIMPLE STRUCTURES

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## DEFINITION OF OPTIMAL FINITE ELEMENT SIZE

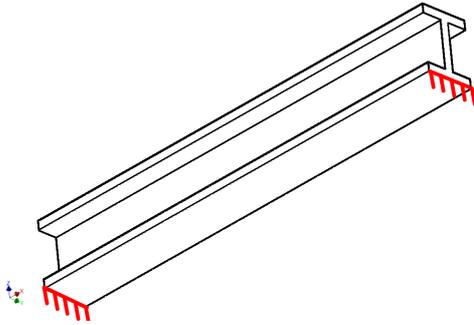
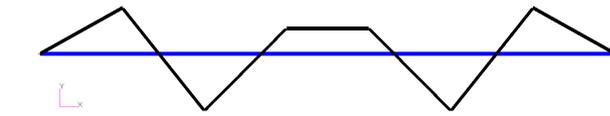
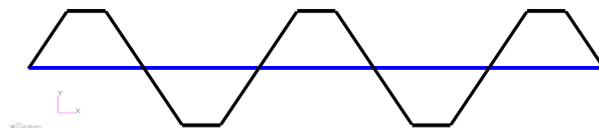


Figure 2. Simply supported beam

$$\text{Discrepancy, } \Delta = \frac{\text{numerical} - \text{analytical}}{\text{analytical}} \quad (1)$$



a). 7 elements,  $\Delta = -6.09\%$



b). 15 elements,  $\Delta = -1.16\%$

Figure 3. Mode shape with 5 half-waves

# SIMPLE REAL CASE

# SIMPLE REAL CASE

## FREQUENCY RESPONSE ANALYSIS PROCEDURE

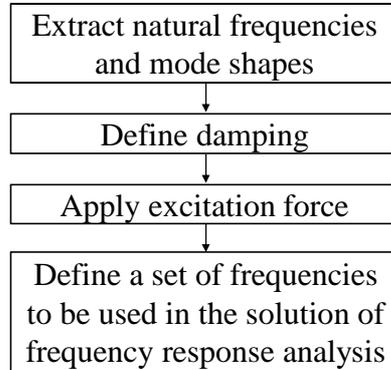


Figure 4. Flowchart of frequency response analysis

# SIMPLE REAL CASE

## MODAL ANALYSIS

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Fringe: modal analysis, A3:Mode 1 : Freq. = 9.0988, Eigenvectors, Translational, Magnitude, (NON-LAYERED)  
 Deform: modal analysis, A3:Mode 1 : Freq. = 9.0988, Eigenvectors, Translational.

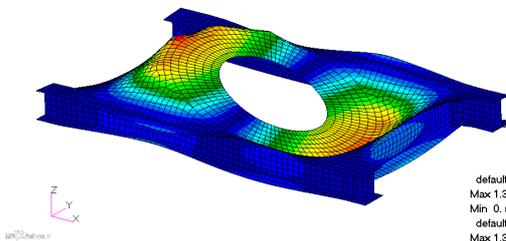


Figure 5. Mode shape 1 of E-motor assembly

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Fringe: modal analysis, A3:Mode 2 : Freq. = 13.395, Eigenvectors, Translational, Magnitude, (NON-LAYERED)  
 Deform: modal analysis, A3:Mode 2 : Freq. = 13.395, Eigenvectors, Translational.

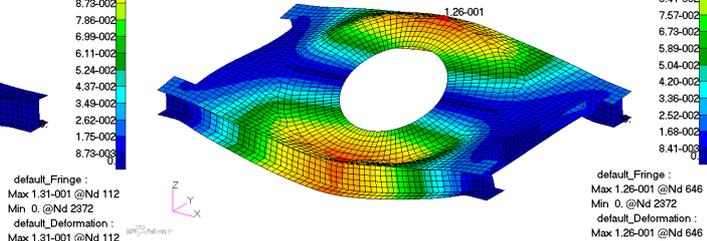


Figure 6. Mode shape 2 of E-motor assembly

# SIMPLE REAL CASE

## FREQUENCY RESPONSE ANALYSIS PROCEDURE

- Modal damping ratio  $\zeta = 0.02$  is suggest for all modes to be conservative.
- The magnitude of unbalance excitation force:

$$F_{unb} = me\omega^2 \quad (2)$$

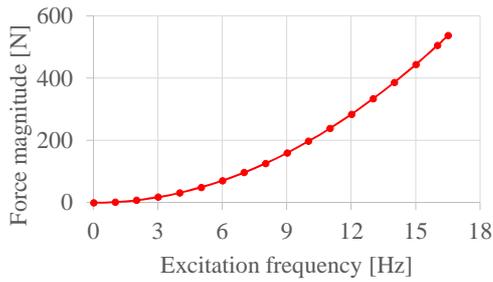


Figure 7. Force magnitude

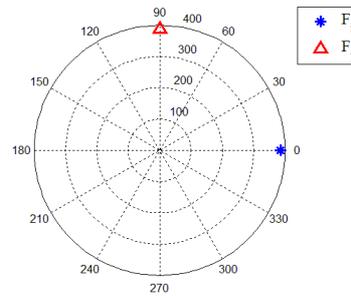


Figure 8. Decomposed force for main continuous rate

# SIMPLE REAL CASE

## FREQUENCY RESPONSE ANALYSIS. NUMERICAL RESULTS

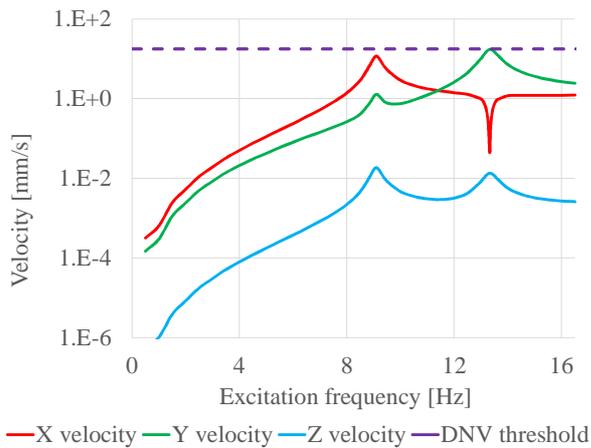


Figure 9. Velocity at COG of E-motor

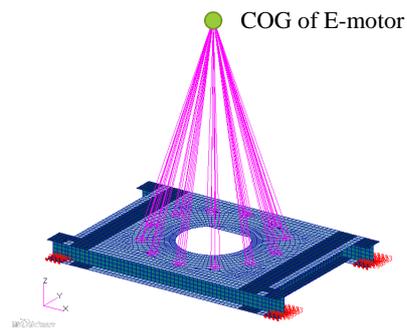


Figure 10. Node for velocity extraction

**Conclusion:**  
Investigation into vibration isolation is reasonable.

# SIMPLE REAL CASE

## VIBRATION ISOLATION

The problem is solved by so-called force transmissibility:

$$T.R. = \frac{F_T}{F_0} = \sqrt{\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2}} \quad (3)$$

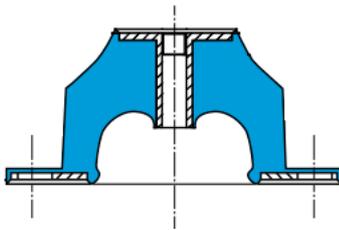


Figure 11. Rubber chock

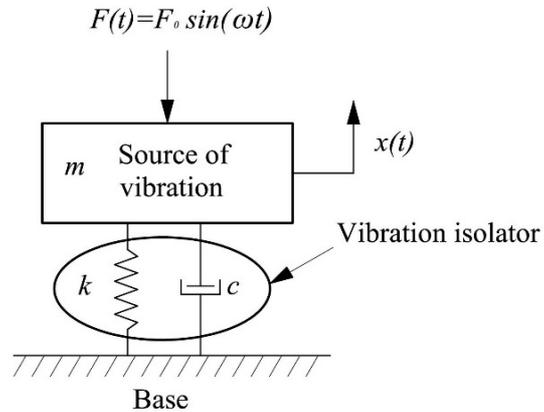


Figure 12. Model of vibration isolation

# SIMPLE REAL CASE

## VIBRATION ISOLATION

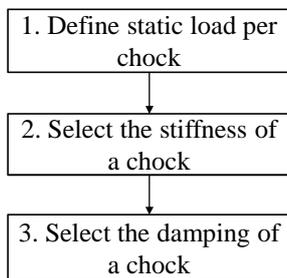


Figure 13. Flowchart of the design of a vibration isolation system

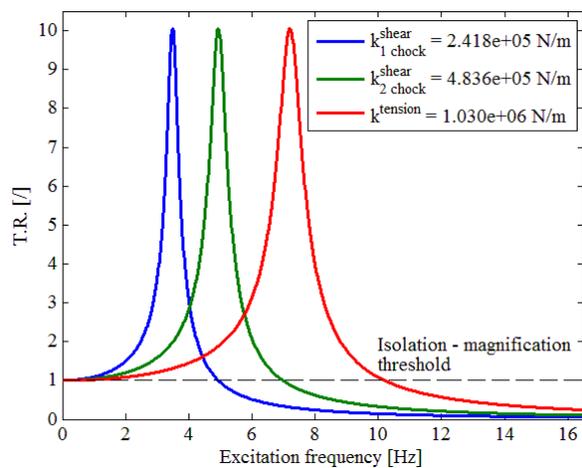


Figure 14. Transmissibility ratio

# SIMPLE REAL CASE

## VIBRATION ISOLATION

3. Select the damping of a chock witch provides the desired isolation.

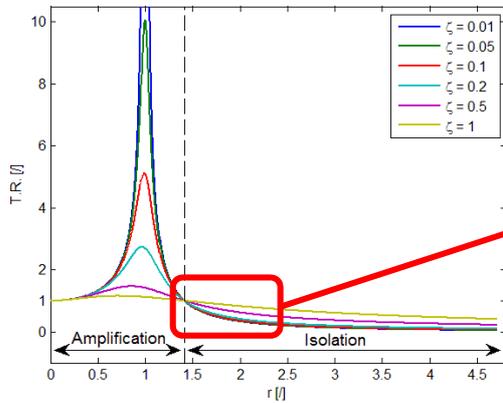


Figure 15. Transmissibility ratio as a function of  $r$  ratio

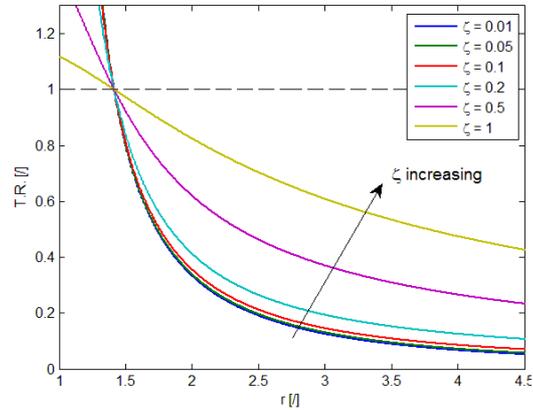


Figure 16. The magnification of isolation area

# SIMPLE REAL CASE

## NUMERICAL RESULTS

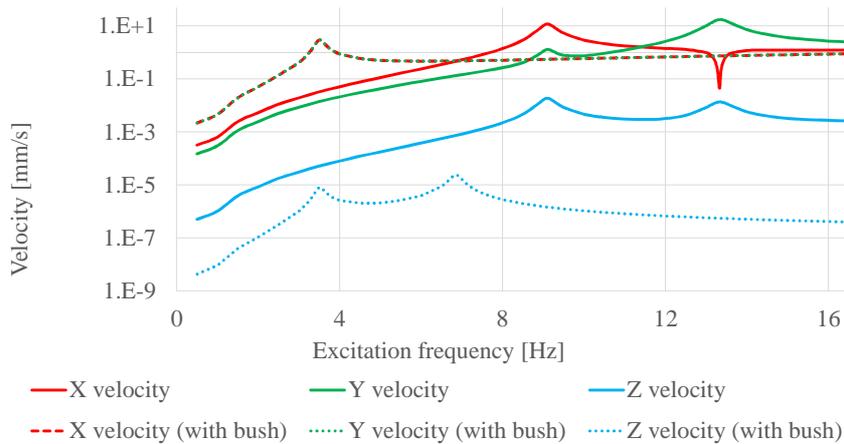


Figure 17. Velocity with and without chocks

**Conclusion:**

- Resonance frequencies are shifted;
- Velocity with chocks are way below threshold.

# COMPLEX REAL CASE

## COMPLEX REAL CASE

### RETRACTABLE AZIMUTH THRUSTER



Figure 18. Retractable azimuth thruster

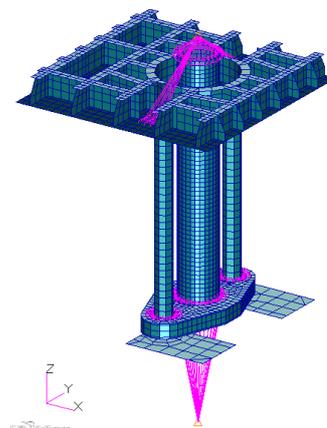


Figure 19. FE modelling

**Conclusion:**  
Vibration influences structural design in the vicinity of the azimuth thruster.

## MODAL ANALYSIS

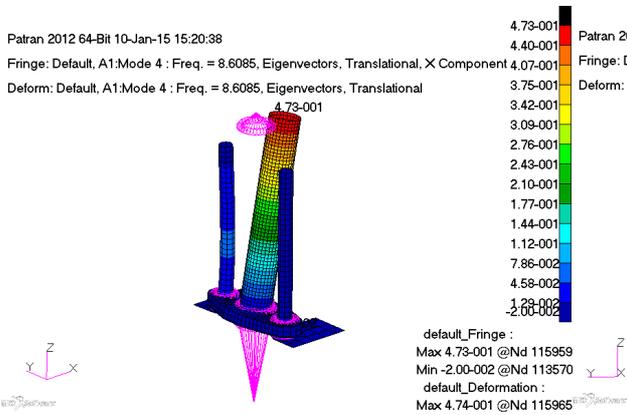


Figure 20. Pendulum mode of stemsection about y-axis

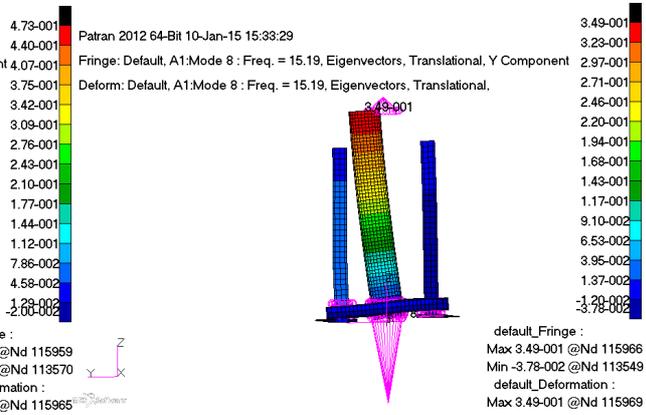


Figure 21. Pendulum mode of stemsection about x-axis

## MODAL ANALYSIS

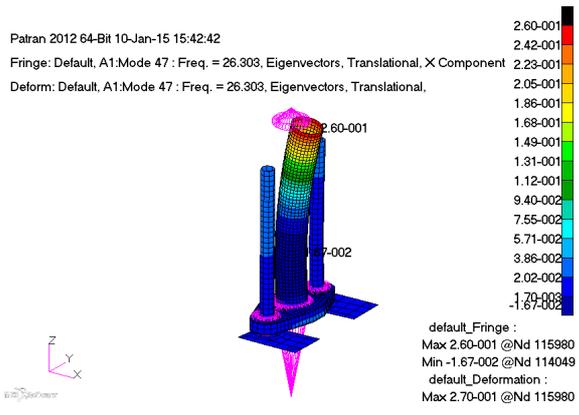


Figure 22. Bending mode of stemsection about y-axis

## DYNAMIC LOADING

- Only  $F_{x.dyn}$ ,  $F_{y.dyn}$ ,  $F_{z.dyn}$ ,  $M_{x.dyn}$  are used.
- A modal damping of 2% is considered.

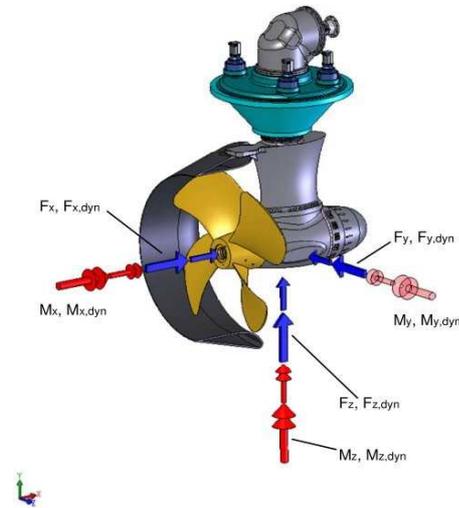


Figure 23. Main static and dynamic forces

## NUMERICAL RESULTS

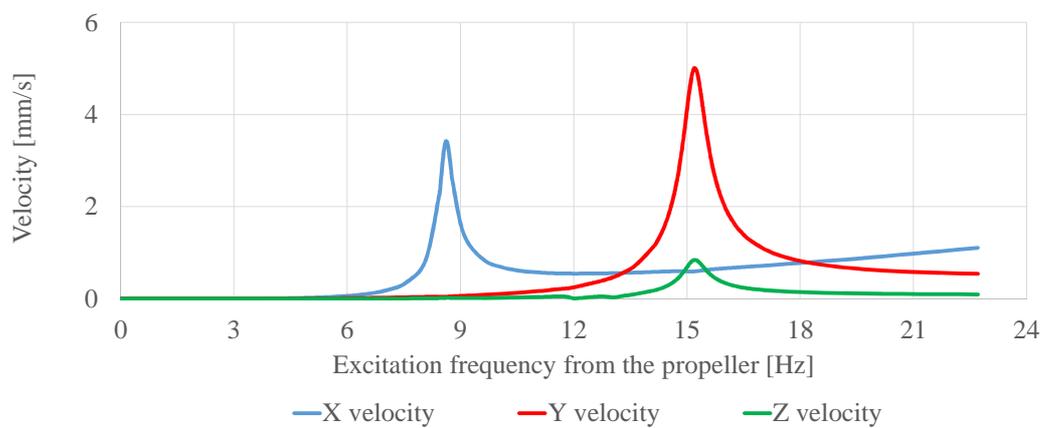


Figure 34. Velocity of COG of gearbox foundation

# CONCLUSION AND FUTURE PROSPECTS

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## CONCLUSION

- The flowchart of vibration analysis and the instructions for its implementation are given;
- The suitability of analysed analytical solutions for the definition of optimal FE size is investigated;
- Required sensitivity studies are performed;
- Modal analysis as a means to identify resonance beforehand is described;
- Static FE model is adapted for the purpose of vibration analysis;
- Local and global vibration responses are obtained;
- Vibration isolation system is designed.

# FUTURE PROSPECTS

- Specify the local response of azimuth thruster with the precise mean values of loading as the functions of steering angle and ship speed;
- Consider hydrodynamic added mass for the submerged surfaces.

# THANK YOU FOR YOUR ATTENTION

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